



A

SECURITY CLASSIFICATION OF THIS PAGE (When Date Entered)

REPORT DOCUMENTATION PA	AGE	READ INSTRUCTIONS BEFORE COMPLETING FORM
1	D-11026	3 RECIPIENT'S CATALOG NUMBER
BOTOR FRAGMENT PROTECTION PROGRAM: S'AIRCRAFT GAS TURBINE ENGINE ROTOR FA OCCURRED IN U. S. COMMERCIAL AVIATIO	TATISTICS ON ILURES THAT	Final Report, 1976—1977.
R. A. DeLucia and J. T. Salvino		8. CONTRACT OR GRANT NUMBER(a)
PERFORMING ORGANIZATION NAME AND ADDRESS Commanding Officer Naval Air Propulsion Center (PE42) Trenton, New Jersey 08628		10. PROGRAM ELEMENT PROJECT, TASK AREA & WORK UNIT NUMBERS None
11 CONTROLLING OFFICE NAME AND ADDRESS National Aeronautics and Space Admin Lewis Research Center Cleveland, Ohio 44135	istration	12. REPORT DATE July-1979  13. NUMBER OF PAGES 29
14 MONITORING AGENCY NAME & ADDRESS(If different to	rom Cantrolling Office)	15. SECURITY CLASS. (of this report)  UNCLASSIFIED  15a. DECLASSIFICATION/DOWNGRADING SCHEOULE
16 DISTRIBUTION STATEMENT (of this Report) APPROVED FOR PUBLIC RELEASE: DISTRI	BUTION UNLIMITE	DTIC

DTIC SELECTE AUG 7 1981

17 DISTRIBUTION STATEMENT (of the abstract entered in Block 20, If different from Report)

A

18. SUPPLEMENTARY NOTES

"Structures and Mechanical Technologies Division, Lewis Research Center, Cleveland, Ohio 44135, Advisor, C. C. Ghamis Other report designation: NASA-CR-165388

19. KEY WORDS (Continue on reverse side if necessary and identify by block number)

Air Transportation and Safety

Aircraft Hazards

Aircraft Safety

Gas Turbine Engine Rotor Failures

20. ABSTRACT (Continue on reverse side if necessary and identify by block number)

This report presents statistical information relating to the number of gas turbine engine rotor failures which occurred during 1977 in commercial aviation service use. The predominant failure involved blade fragments, 89 percent of which were contained. Although fewer rotor rim, disk, and seal failures occurred, 50%, 100% and 33% respectively were uncontained. Seventy-two percent of the 163 rotor failures occurred during the takeoff and climb stages of flight.

DD 1 JAN 73 1473

EDITION OF 1 NOV 68 IS OBSOLETE S/N 0102-014-6601 :

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE (When Date Entered)

٠ ک

ئے	UHITY CLASSIFICATION OF THIS PAGE(When Data Entered)	_
1		1
1		ł
1		
1	•	
1		
1		1
		1
		1
1		-
1		1
1		1
1		1
		1
		1
		Ì
1		l
l		1
-		1
1		1
1		۱
1		1
1		1
1		
ł		
1		1
1		١
1		
j		1
1		١
ı		١
1		1
-		1
-		1
1		1
		-
ł		

## NAVAL AIR PROPULSION CENTER TRENTON, NEW JERSEY 08628

PROPULSION TECHNOLOGY AND PROJECT ENGINEERING DEPARTMENT

NAPC-PE-23 NASA-CR-165388 JULY 1979

ROTOR FRAGMENT PROTECTION PROGRAM: STATISTICS ON AIRCRAFT

GAS TURBINE ENGINE ROTOR FAILURES THAT OCCURRED IN U. S.

COMMERCIAL AVIATION DURING 1977

Prepared by:

R. A. DeLUCIA

Approved by:

R. C. BONDI,

Commander, USN Director, PE

APPROVED FOR PUBLIC RELEASE: DISTRIBUTION UNLIMITED

AUTHORIZATION: NASA DPR C-41581-B, MOD.8

Accession For

NTIS CHARI
DESCRIBE TO
Unsubscinedd TO
J differtion

Proteibation/
Availabaity Codes
Avail ind/or
Joist Special

## ACKNOWLEDGEMENTS

We thank the Flight Standards National Field Office, Federal Aviation Administration, Oklahoma City, Oklahoma, for their cooperative effort in providing the basic data used for this report.

## TABLE OF CONTENTS

	Page
REPORT DOCUMENTATION PAGE DD Form 1473	
TITLE PAGE	
ACKNOWLEDGEMENTS	i
TABLE OF CONTENTS	ii
LIST OF FIGURES	iii
INTRODUCTION	1
RESULTS	1-3
CONCLUSIONS	3
Figures 1 through 7	4-10
APPENDIX A	<b>A-1</b> to A-14
DISTRIBUTION LIST	Inside rear cover

## LIST OF FIGURES

Figure No.	<u>Title</u>	Page
1	Incidence of Rotor Failure in U. S. Commercial Aviation - 1977	4
2	Component and Fragment Type Distribution for Contained and Uncontained Rotor Failures - 1977	5
3	The Incidence of Rotor Failure in U. S. Commercial Aviation According to Engine Type Affected - 1977	6
4	Rotor Failure Cause Categories - 1977	7
5	Flight Condition at Rotor Failure - 1977	8
6	Uncontained Rotor Failure Distributions According to Cause and Flight Condition - 1976 - 1977	9
7	The Incidence of Uncontained Rotor Failure in U.S. Commercial Aviation - 1962 - 1977	10

## INTRODUCTION

This report has been prepared as part of the Rotor Fragment Protection Program (RFPP), which is sponsored by the National Aeronautics and Space Administration (NASA)<sup>1</sup> and conducted by the Naval Air Propulsion Center (NAPC). The objective of the RFPP is to develop criteria for the design of devices that will be used on aircraft to protect occupants and the aircraft structure from the potentially lethal and devastating fragments that are generated by gas turbine engine rotor failures.

Presented in this report are statistics on gas turbine rotor failures that have occurred in U. S. commercial aviation during 1977. These statistics are based on data compiled from the Flight Standards Service Difficulty Reports (SDRs) that were published by the Department of Transportation, Federal Aviation Administration (FAA). The compiled data were analyzed to establish:

- 1. The incidence of rotor failures and the incidence of contained and uncontained  $^2$  rotor fragments.
- 2. The distribution of rotor failures with respect to engine rotor component; i.e., fan, compressor or turbine rotors and their rotating attachments or appendages such as spacers and seals.
- The type of rotor fragment (disk, rim or blade) typically generated at failure.
  - 4. The cause of failure.
  - 5. The type of engines involved.
  - 6. The flight condition at the time of failure.

## RESULTS

- 1. The data used for analysis are contained in APPENDIX A. The results of these analyses are shown in Figures 1 through 7.
- a. Figure 1 shows that 163 rotor failures occurred in 1977. These rotor failures accounted for approximately 7.6% of the 2132 shutdowns experienced by the gas turbine powered U. S. commercial aircraft fleet during 1977. Rotor fragments were generated in 102 of the failures

<sup>&</sup>lt;sup>1</sup>NASA DPR C-41581-B, Mod. 8.

<sup>&</sup>lt;sup>2</sup>An uncontained rotor failure is defined as a rotor failure that produces fragments which penetrate and escape the confines of the engine casing.

experienced and, of these, 15 (14.7% of the fragment producing failures) were uncontained. This represents an uncontained failure rate of 2.3 per million gas turbine engine powered aircraft flight hours, or 1.3 per million engine operation hours. Approximately 6.5 million and 19.7 million aircraft flight and engine operating hours, respectively, were logged by the U. S. commercial aviation fleet in 1977.

- b. Figure 2 shows the distribution of rotor failures that produced fragments according to the engine component involved -- fan, compressor, turbine; the types of fragments that were generated; and the percentage of uncontained failures according to the type fragment generated. These data indicate that:
- (1) The incidence of turbine rotor fragment producing failures was approximately two and one-half times greater than that of compressor rotor fragment producing failures; these corresponded to 65.7% and 26.5%, respectively, of the total number of rotor failures. Fan rotor failures accounted for 7.8% of the fragment producing failures experienced.
- (2) Blade fragments were generated in 91.2% of the rotor failures; 10.8% of these were uncontained. The remaining rotor fragments failures (8.8%) produced disk, rim and seal fragments, of which 100%, 50% and 33.3%, respectively were uncontained.
- c. Figure 3 shows the rotor failure distribution among the types of engines that were affected, and the total number of that type engine in use.
- d. Figure 4 shows what caused the rotor failures to occur. Of the known causes of failure (1), the dominant causal factors were: (1) Secondary Causes (36.8%); (2) Foreign Object Damage (34.7%); and (3) Design and Life Prediction Problems (25.3%).
- e. Figure 5 indicates the flight conditions that existed when the various rotor failures occurred. Approximately 72% of the 163 rotor failures occurred during the takeoff and climb stages of flight. Approximately 75% of the rotor fragment producing failures, and 87% of the uncontained rotor failures, occurred during these same stages of flight. The highest percentage of uncontained rotor failures (60%) were experienced during takeoff.
- f. Figure 6 is a new cumulative tabulation that describes the distribution of uncontained rotor failures according to fragment type, engine component involved, cause category and flight condition<sup>(2)</sup> for the years 1976 and 1977. This figure will be expanded yearly to include all subsequent uncontained rotor failures. These data indicate that:

<sup>(1)</sup> Because of the high percentage of unknown causes of rotor failure, the percentages were based on the total number of known causes.

<sup>(2)</sup> Takeoff and climb are defined as "High Power", all other conditions are defined as "Low Power".

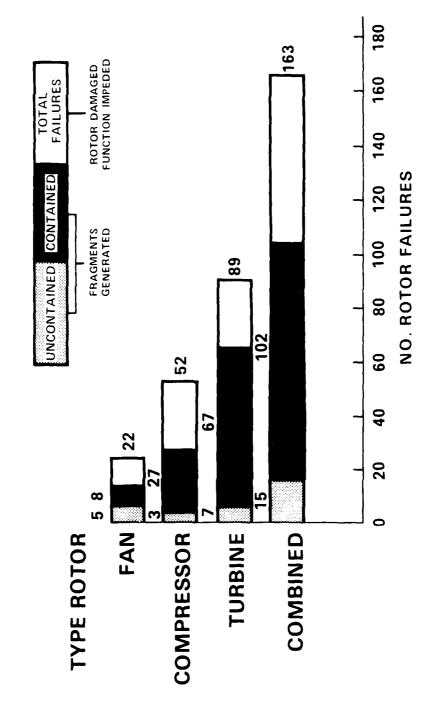
for "design and life prediction problems" the numbers of uncontained failures were about equal between "high" and "low" power (namely 5 and 4); but for all other causes, the prevailing condition was "high power". Additional conclusions should become evident from this table with the accumulation of future data.

g. Figure 7 shows the annual incidence of uncontained rotor failures in commercial aviation for the years 1962 through 1977. During 1977, the incidence of uncontained rotor failure remained constant over the previous year, 1976. Over the past four years, 1974 through 1977, an average of 16 uncontained rotor failures per year have occurred. During this same time period, the rate of uncontained rotor failures has remained relatively constant at an average of approximately 1 per million engine operating hours.

## CONCLUSIONS

- 1. The incidence of rotor failure and uncontained failure is significantly high enough to warrant continuation of the experimental and analytical efforts that constitute the Rotor Fragment Protection Program.
- 2. Of all the types of fragments generated at rotor failure, disk and fan blade fragments, because of their size, high energy content and high rate of uncontainment, continue to be the threat that must be addressed in the RFPP.
- 3. It appears that causes beyond the control or scope of present technology such as FOD, structural life and integrity prediction, and secondary effects, are still primarily responsible for most of the rotor failures that occur.

## IN U.S. COMMERCIAL AVIATION 1977 INCIDENCE OF ROTOR FAILURE

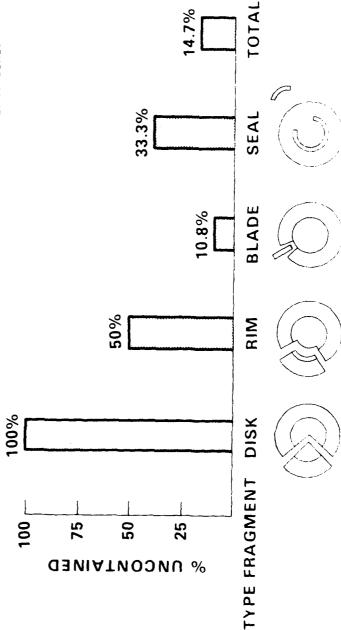


## CONTAINED AND UNCONTAINED ROTOR FAILURES(1) -1977 COMPONENT AND FRAGMENT TYPE DISTRIBUTIONS FOR

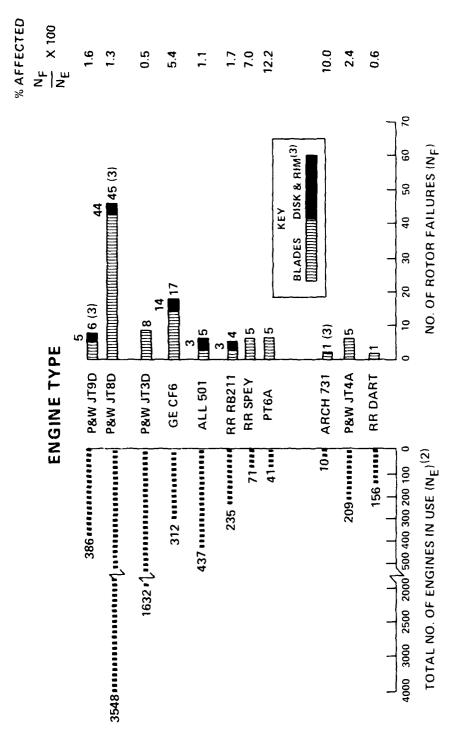
FNGINE		TYP	E OF	FRAGN	MENT (	TYPE OF FRAGMENT GENERATED	ATED			
ROTOR	ā	DISK	8	RIM	BL.	BLADE	SE	SEAL	101	TOTALS
COMPONENT	T.	TF UCF		TF UCF	41	UCF	1 F	TF UCF	TF	UCF
FAN	0	0	0	0	8	5	0	0	8	5
COMPRESSOR	0	0	2	2	24	-	-	0	27	m
TURBINE	2	2	2	0	61	4	2	-	67	_
TOTALS	2	2	4	2	93	10	3	-	102	15

(1) FAILURES THAT PRODUCED FRAGMENTS

TF - TOTAL FAILURES UCF -- UNCONTAINED FAILURES

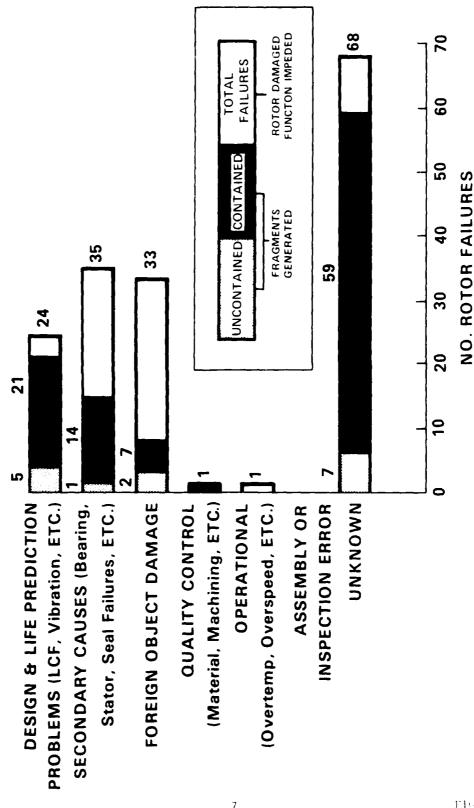


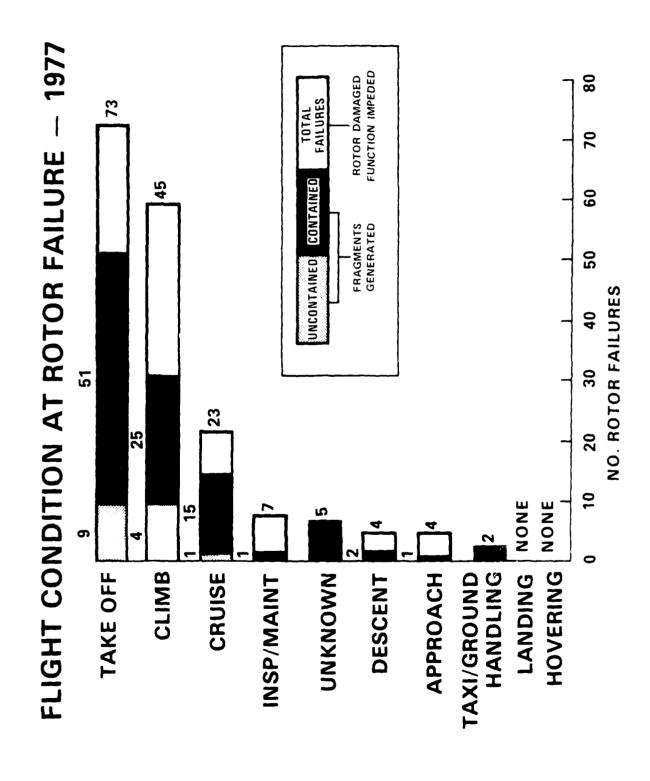
## THE INCIDENCE OF ROTOR FAILURE(1) IN U.S. COMMERCIAL AVIATION ACCORDING TO ENGINE TYPE AFFECTED - 1977



NOTES: (1) FAILURES THAT PRODUCED FRAGMENTS
(2) YEARLY AVG. OF AIRCRAFT IN USE AT END OF EACH MONTH
(3) 1 SEAL-SPACER FAILURE INCLUDED IN DISK/RIM COMPILATION

## ROTOR FAILURE CAUSE CATEGORIES - 1977



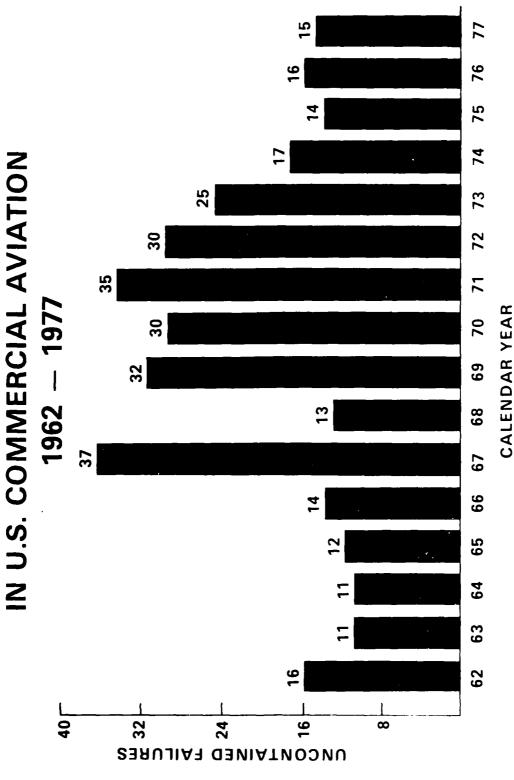


## UNCONTAINED ROTOR FAILURE DISTRIBUTIONS ACCORDING TO CAUSE AND FLIGHT CONDITION(1) 1976-1977

TYPE OF FRAGMENT	ENGINE	DESIGN & LIFE	DESIGN & LIFE PRED PROBLEMS	SECONDARY CAUSES	DARY	FOREIGN DAMAGE		QUALITY CONTROL	17 Y ROL	UNKNOWN	IOWN	SUBT	SUBTOTALS	
GENERATED	COMPONENT	HIGH POWER	LOW POWER	HIGH	LOW	HIGH POWER	LOW	HIGH	LOW	HIGH	LOW	HIGH POWER	LOW	TOTALS
	FAN											0		
DISK	COMPRESSOR			,			}					0	0	
,	TURBINE		2									0	2	2
	FAN										}	0	0	
RIM	COMPRESSOR	2								2		4	0	
	TURBINE											0	0	4
	FAN	3	-			3		2				30	-	
BLADE	COMPRESSOR			2						က		5	0	
	TURBINE			က						2	-	5	2	21
	FAN											٥	0	
SEAL	COMPRESSOR										-	0	-	
	TURBINE			2						-		3	0	4
SUBTOTALS	TALS	ç	4	7	0	3	0	2	0	8	2	25	9	
TOTALS	S	,   	6	7		3		2		10	0	! 		31

1) TAKEOFF AND CLIMB ARE DEFINED AS "HIGH POWER" AND ALL OTHER CONDITIONS ARE DEFINED AS "LOW POWER

# THE INCIDENCE OF UNCONTAINED ROTOR FAILURES



## APPENDIX A

Data on Rotor Failures in U. S. Commercial Aviation for 1977. Compiled from the Federal Aviation Administration Service Difficulty Reports.

## DATA COMPILATION KEY:

## Component Code:

- F Fan
- C Compressor
- T Turbine

## Fragment Type Code:

- D Disk
- R Rim
- B Blade
- S Seal
- N None

## Cause Code:

- 1 Design and Life Prediction Problems
- 2 Secondary Causes
- 3 Foreign Object Damage
- 4 Quality Control
- 5 Operational
- 6 Assembly and Inspection Error
- 7 Unknown

## Containment Condition Code:

- C Contained
- NC Not Contained
- N No Fragments Generated

## Flight Condition Code:

- l Insp/Maint
- 2 Taxi/Grnd Hdl
- 3 Takeoff
- 4 Climb
- 5 Cruise
- 6 Descent
- 7 Approach
- 8 Landing
- 9 Hovering
- 10 Unknown

CHARACTERISTICS OF ROTOR FAILURES - 1977

SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	FRAGMENT	CAUSE	CONTAINMENT	FLIGHT
01147035	1/2	NAL	B727	JT8D	£+	B	7	C	3
01177033	1/1	NAL	B727	JT8D	T	В	7	S	3
01207032	1/5	AAA	DC9	JT8D	S	В	7	S	3
01207033	1/1	AAA	BA111	909	T	В	7	O O	3
01217035	1/6	VAL	B727	JT8D	Įτι	В	7	O	3
01247032	1/7	SWAX	B737	JT8D	Ţ	В	7	S	9
02147018	2/3	TWA	B747	JT9D	T	В	2	J	5
02117022	2/6	TIAS	DC10	CF6	IJ	В	3	O O	3
02227032	2/4	AAA	DC9	JT8D	I	В	1	S	3
02227033	2/7	OZA	DC9	JT8D	Į±,	В	1	NC	3
02247037	2/13	EAL	L1011	RB211	S	В	2	S	5
02257036	2/3	RDLS	DC8	JT3D	Ţ	В	7	NC	5
03017031	2/11	CAPS	DC8	JT4A	T	В	7	C	-3
03027020	2/18	BNF	B727	JT8D	Ţ	В	2	NC	~
03097023	2/19	PAA	8747	JT9D	Ĺ	8	~	N(	<b>~</b> †

NAPC -PE-23

CHARACTERISTICS OF ROTOR FAILURES - 1977 (Continued)

SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	FRAGMENT	CAUSE	CONTAINMENT	FLIGHT
02027036	1/26	0ZA	DC9	JT8D	T	В	~	S	3
02017037	1/22	EAL	L1011	RB211	S	В	2	S	3
03107027	2/17	PSAX	B727	JT8D	ſz.,	В	7	C	10
03107028	2/23	WAA	B737	JI8D	S	æ	2	S	3
03187027	3/9	NCA	CV580	501	1	D	7	NC	3
03227024	3/13	TWA	B707	JT3D	S	В	7	ပ	7
03237025	3/14	0ZA	DC9	JT8D	T	В		S	3
03297022	3/14	WAA	B737	JT8D	ပ	В	2	C	3
04047027	3/28	TXI	DC9	JT8D	S	В	7	J	2
04117024	3/29	PAA	B747	JT9D	H	В	7	S	7
04127024	4/2	TWA	B707	JI3D	Ħ	В	2	S	7
04187026	3/28	RAIX	DC8	JT4A	C	В	2	S	2
04207022	7/7	VAL	DC10	CF6	L	В	1	C	10
04257022	4/12	TWA	B727	JT8D	H	В	7	S	5
04207022	4/20	VAL	DC10	CF6	Н	В	7	C	5
04297025	4/16	NWA	DC10	JT9D	H	В	_	C	3

CHARACTERISTICS OF ROTOR FAILURES - 1977 (Continued)

						minute of the		minute to the control of	
SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	TYPE	CAUSE	CONDITION	CONDITION
05067024	4/28	NAL	DC10	CF6	<b>.</b>	В	7	U	7
05117019	4/27	FAL	CV580	501	O	В	7	၁	7
05137025	4/11	EAL	DC9	JT8D	T.	8	7	Ú	3
05167022	4/29	TWA	B707	JT4A	т	В	٣	ပ	2
05187022	4/28	AAA	BA111	909	Т	g	1	ပ	3
05237024	5/5	0Z <b>A</b>	DC9	JT8D	o	В	1	၁	3
05277025	5/14	TWA	B727	JT8D	Т	В	7	၁	7
06017024	5/14	EAL	DC9	JT8D	П	æ	1	၁	5
06137027	5/28	AWI	DC9	JTSD	H	æ	7	S	7
06077027	5/23	AAA	6DG	JT8D	Н	В	7	C	3
06087025	5/25	ONAS	DC8	JT3D	Т	æ	2	S	7
06167023	6/1	VAL	DC10	CF6	S	æ	7	NC	7
06207024	5/25	BCAT	L35	TFE731	Т	S	7	C	5
06277024	6/12	PSAX	B727	JT8D	၁	В	7	C	3
06277025	6/14	EAL	6DC	JT8D	Ц	В	7	၁	3
07147017	1/1	TWA	B707	JT3D	S	В	1~	၁	<u>ش</u>

NAPC-PE-23

CONDITION FLIGHT CONTAINMENT CONDITION S S SC NC NC ပ ပ ပ ပ CHARACTERISTICS OF ROTOR FAILURES - 1977 (Continued) CAUSE FRAGMENT TYPE 24 S COMPONENT ပ ENGINE RB211 RB211 JT8D JT4A PT6A JT8D JT8D JT3DJT8D JT9D JT8D CF6 CF6 CF6 501 AIRCRAFT L1011 ND262 CV580 L1011 RAIII DC10 DC10 B737 DC10 B707 B747 DC9 DC9 DC9 DC8 DC9 SUBMITTER CAPS VAL AWI TWA AAL AAA FAL TXI FAL 0ZA **AA**L AAA VAL DAL TWA 1.1.4 DATE 6/28 7/15 7/29 7/12 7/28 7/23 7/29 7/28 7/11 6// 8/1 8/5 8/7 8/1 08017020 08047019 08127025 08167019 08227023 09157023 07197024 07277026 08097017 08127024 08297025 09027022 9616161 07287021 09077024 08047021 SDR NO.

CHARACTERISTICS OF ROTOR FAILURES - 1977 (Continued)

SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	FRAGMENT	CAUSE	CONTAINMENT	FLIGHT
09227029	9/2	PSAX	B727	JT8D	Ę	В	7	၁	7
09267031	9/6	TWA	B727	JT8D	Ħ	В	7	S	3
09277022	8/29	WAA	B737	JT8D	၁	В	7	C	7
09287026	9/13	NAL	DC10	CF6	H	82	7	S	3
10067027	9/16	AAA	BAlll	506	H	В	7	S	3
10117025	9/6	VAL	DC10	CF6	H	84	7	ပ	3
10187024	9/22	CAL	DC10	CF6	H	g.	7	S	7
10197024	5/6	CAIT	SD330	PT6A	H	В	1	v	9
10217023	9/29	<b>9</b> Z0	DC9	JT8D	ن	S	7	S	10
10267022	10/10	NAL	DC10	CF6	Ú	В	1	O)	3
10287023	9/01	DAL	DC8	JT3D	[24	В	7	O	3
10287024	10/7	TWA	B727	JT8D	S	23	2	S	7
11027024	10/17	VAL	B747	JT9D	[±.	23	3	NC	3
11037027	10/14 AAA	AAA	DC9	JT8D	H	В	7	C	3
11117023	9/01	VAL	B727	JT8D	Ú	Ж	<b>,</b> .	Ú	~
11177022	10/29 ANE	ANE	DHC6	PT6A	3	22	^;	O	

CHARACHIRESTESS OF REFORE FATICEES = 1977 to attaced)

NAPC+11-13

3	<u>:</u>					PRACMINI			
SDR NO.	DATE.	SUBMITTER	AIKCKAFI		17470.000		) (1) (1) (1) (1) (1) (1) (1) (1) (1) (1		
7702771	17/11	WAL.	0010	CF6	1	20	-	DT:	. •
11217023	19722 CM	CALT	क्ष हत्तर	Plea	<i></i>	Σĵ.		()	71
11287021	10/30 AM	AVA	600	1180	1.	w)	! ~	3)	~~
11307021	10/39 CAF	CAIT	SD330	PT6A	; <b>-</b>	z:	~	C	c
12017026	H/15 TWA	TWA	87.27	<u> </u>		æ	1 /	.)	~
61027021	11/28 FM	FML	08570	501			-	D!:	·u
12197020	12/3	TKA	8727	.17.830		భా	- 1	-	. •
12217015	12/5	CAL	0100	910	3	25	^ ;	.)	
12217020	108 / E	CAPS	90.8	T.S.		an		J::	*
91108029	12/3	7777	DC9	185		at.		_	. •
01128030	12/22 AW	TWV	900	ITSD	_	st	٠.	J	44
01188031	127 H 3GA	Y.C.Y	(1961)	21.85	_	s#		J-0	
11198150	L. CO HAL	HAL.	$t_{i}$ $\mu$			v.		-	
e providiti		177	F Ja	7.66 2.	P on	-		• •	
	* **								
		į,							

ARACTERISTICS OF ROTOR FAILURES - 1977 (Continued)

		5	AKACTERISTI	CS OF RUI	CHARACTERISTICS OF ROIOR FAILURES - 1977 (Continued)	- 197/ (Cor	itinued)		
SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	FRAGMENT	CAUSE	CONTAINMENT	FLIGHT
02187095	2/18	NAL	DC10	CF6	Ţ	S	7	ပ	10
01057027	1/5	NCA	CV580	501	£-1	g	7	C	5
01127031	1/12	BNF	DC8	JT3D	T	83	7	S	7
03257024	3/25	TWA	B727	JT8D	Т	В	7	S	5
01277034	1/27	TXI	STC24D	DART	Ħ	В	7	S	2
02027038	2/2	AAA	BA111	506	H	В	3	O	3
02177022	2/17	TIAS	DC10	CF6	၁	В	6	C	3
01217011	1/12	PAI	B737	JT8D	Ĺt-	z	3	z	3
01257034	1/11	NAL	B727	JT8D	H	z	7	z	• 7
01277030	1,4	SRAX	L382	501	၁	z	3	z	1
03037013	2/20	SBWS	DC8	JT3D	Ĭτι	z	3	z	3
03037024	2/16	HAJ	L188	501	S	z	3	z	1
03077027	2/23	TWA	B747	JT9D	ĬΤ	z	2	z	-7
01107025	1/10	MIAS	B720	JT3C	၁	z	~	z	3
03177022	3/5	FAL	CV580	501	S	z	3	z	-7
04057023	3/2	FECT	MD20	CF7	Į,	z	~	и	

CHARACTERISTICS OF ROTOR FAILURES - 1977 (Continued)

						JANSING V GSI		PERSONAL PROPERTY.	000 100
SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENCINE	COMPONENT	TYPE	CAUSE	CONDITION	CONDITION
04067027	3/18	AAA	BA111	909	€	z.	~1	×	~
03257024	3/15	TWA	B727	JT8D	S	z	7	z	iC.
04207023	6/7	TWA	B707	JI3D	1	z	<b>?</b> ↓	×	<b>\</b> \$
05067027	4/24	EAL	L1011	RB211	Ĺ	z	3	z	3
05107023	4/26	ACAX	B737	JT8D	S	Z	3	z	3
05177025	4/29	HAL	DC9	JT8D	Ţ	z	2	z	3
05277024	5/19	CAPS	DC8	JT3D	C	z	7	z	<b>7</b> 7
06017026	5/16	NWA	DC10	JT9D	C	z	3	z	7
06027028	5/19	TWA	L1011	RB211	П	z	2	×	7
06067007	5/19	AAA	DC9	JT8D	Įτ	z	3	×	3
06227020	6/1	AAA	ND262	PT6A	T	z	7	×	9
07057026	6/23	TWA	B727	JT8D	C	z	c <del>1</del>	z	.0
07087020	6/25	EAL	9C9	JT8D	Ţ	×	7	v	3
07087021	6/21	AAA	ND262	PT6A	Τ	z	1	×	-
07127018	6/28	TWA	B707	JT30	C	z	~1	z	e#
07157025	47/4	ASA	B727	JT80	<u>-</u>	26	٠.	31	

NAPC+PE-23

CONDITION FLIGHT CONTAINMENT Z  $\mathbf{z}$ Z  $\mathbf{z}$ Z  $\mathbf{z}$ CHARACTERISTICS OF ROTOR FAILURES - 1977 (Continued) CAUSE FRAGMENT TYPE Z Z z  $\mathbf{z}$ 7. 2. COMPONENT  $\circ$ ENGINE RB211 JT8D JT3D JT9D JT3CJT8D JT3D DART CF6 501 CF6 501 909 501 501 JT4 71T AIRCRAFT BA111 L1011 FH227 DC10 DC10 B720 L188 L188 L188 B737 B707 B747 L382 B737 B707 8707 DC8 SUBMITTER SWAX TIAS ACAX SWAX AAIX TWA AAA 0ZA TWA TWA TWA MAL NAL HAL HAL VAL TWA 8/10 DATE 7/.11 7/19 7/30 7/30 7/26 8/12 8/28 67/8 9/6 9/2 7/3 8/3 07227019 08037018 08047020 08107023 08127023 08197025 08227024 08237026 08267022 09137024 09237023 09307022 09267032 09127020 10117026 08037102 08267024 SDR NO.

PASA HELS I SEE ROLDS FAILURES - 1977 (Continued)

						11 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1		THE PERSON	
SDR NO.	DATE	SUBMITTE	AIRCEALL	EX LX	COMPONENT	TYPE	CAUSE	CONDITION	CONDITION
11097017	10/1. AM	NWI.	F.00	1785	i.e.	ue.	~	z	3
11297025	2711	WAI.	8720	9 H	, j	25		z	1
11307018	17/11	TWA	8707	M+ TI.		58	~ 1	×	5
11307019	11/15 TWA	TWA	B707	JT4A	<del></del>	26	τı	<b>7</b> .	٣
12017024	11/12 ACAX	ACAX	L188	501	$\vdash$	24	- 1	×	5
12017025	11/12 TWA	TWA	B707	JT30	Ü	24	~ 1	z	3
12057027	11/17 AAL	YAT	B74.7	JT9D	<b>⊢</b>	×	<i>r</i> ~	z	<b>7</b>
12067014	61/11	PAA	B727	JT8D	íu	25	~	×	4
12137025	11/28	NYA	261L	CT58	<del>[ ·</del>	\$*	~1	z	7
12147013	11/22 EAL	EAL	B727	JT8D	ķi.	2*	~1	z	7
12147025	11/30 VAL	VAI.	B737	JT8])	<u>:</u>	z	~	z	7
12057026	<del></del>	TIAS	1.382	501	Ü	25	~	z	7
01068032	17.16	CAPS	×. ×.	H.4A	_	25	۲,	×	ব
09307022	D# 7.5	W13	B 7.20	11 %		æ	~	z	~
911 1792 :	.i.	MAS	13 53 25	JF 11	·_	24	~	×	~
1670, 2150	11	TW.	; =	1	<u> </u>	11		24	ι <u>ς</u>

NAPC-PE-23

FL IGHT CONDITION CONTAINMENT CONDITION Z z Z CAUSE CHARACTERISTICS OF ROTOR FAILURES - 1977 (Continued) FRAGMENT TYPE Z COMPONENT ENGINE RB211 JT8D 501 AIRCRAFT Unknown L188C DC9 SUBMITTER AFLX TWA DAL DATE 3/20 3/24 3/9 03247114 04017024 03317026 SDR NO.

